

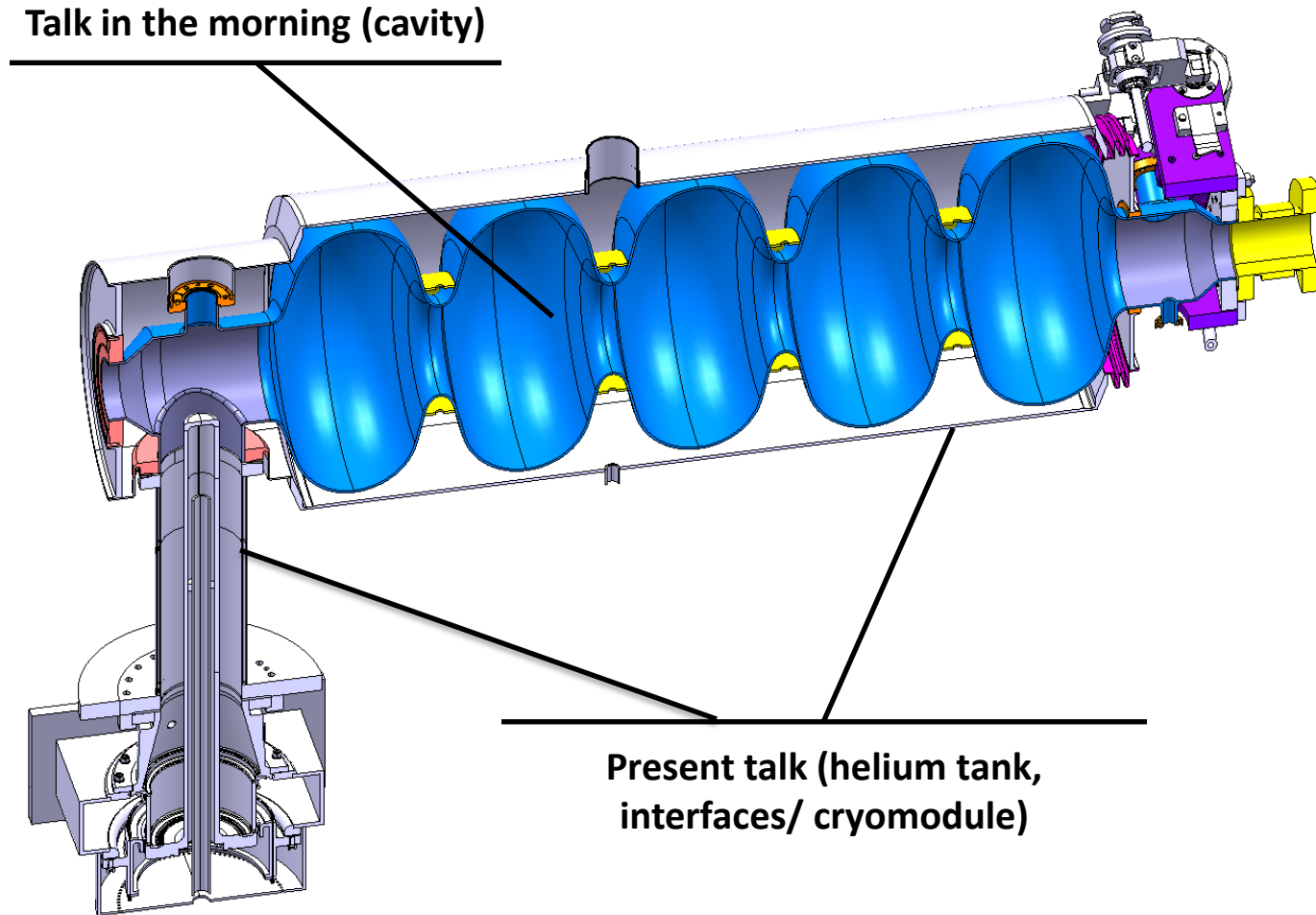


Thermo-mechanical aspects of SPL b=1 cavity helium vessel and cryomodule interface

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+ discussions and contributions from CEA and
CNRS colleagues

Introduction

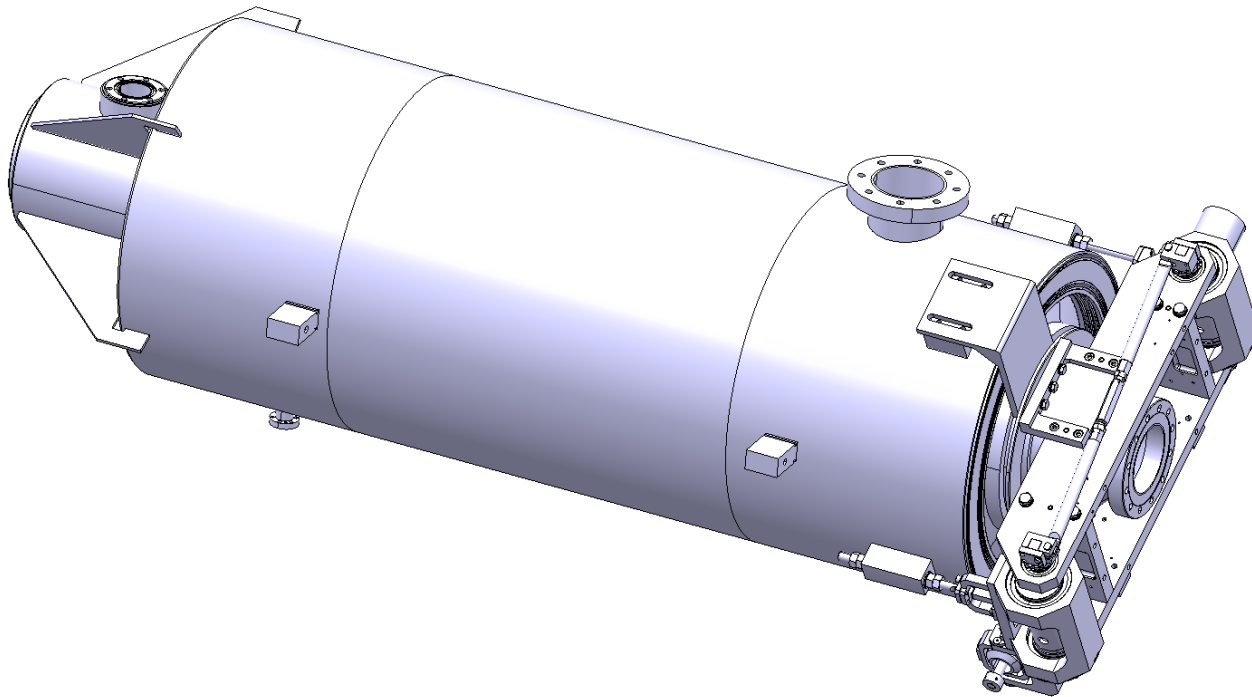
- SPL beta = 1 cavity + helium tank + tuner + main coupler + bellow to next cavity



- Helium tank
- Interface / cryomodule vacuum vessel (main coupler “double wall”)
- Interface / cryomodule cryogenic lines
- Conclusions

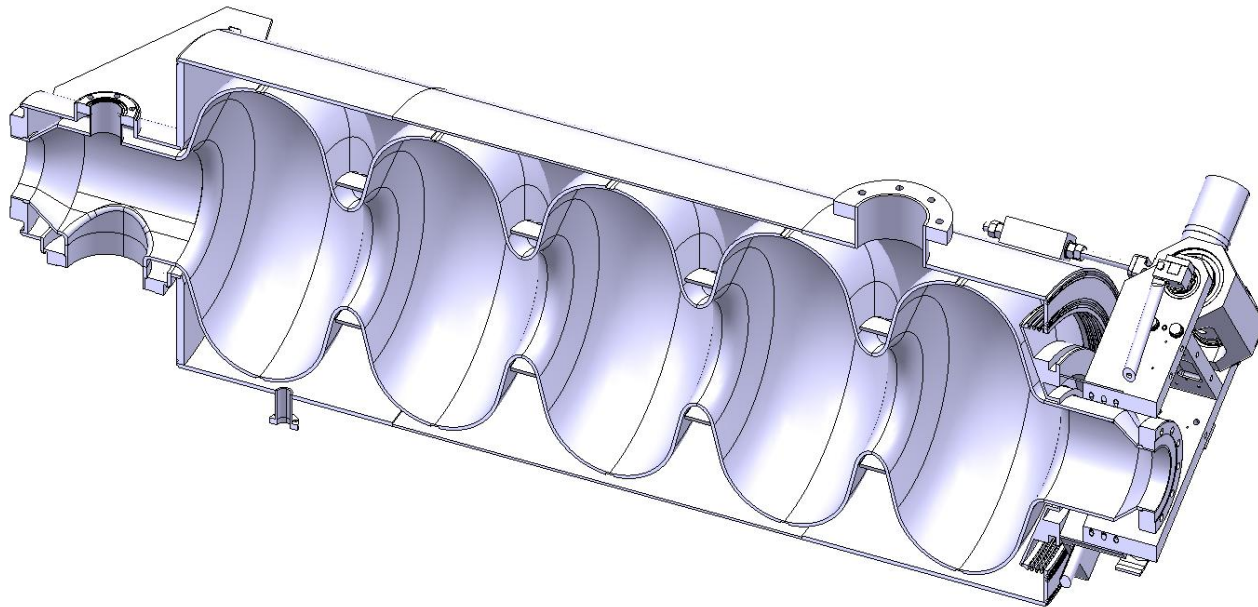
Helium tank

- Helium tank for short cryo-module cavities (first set)
 - The most recent design proposed by CEA – meeting foreseen today for discussion on this subject



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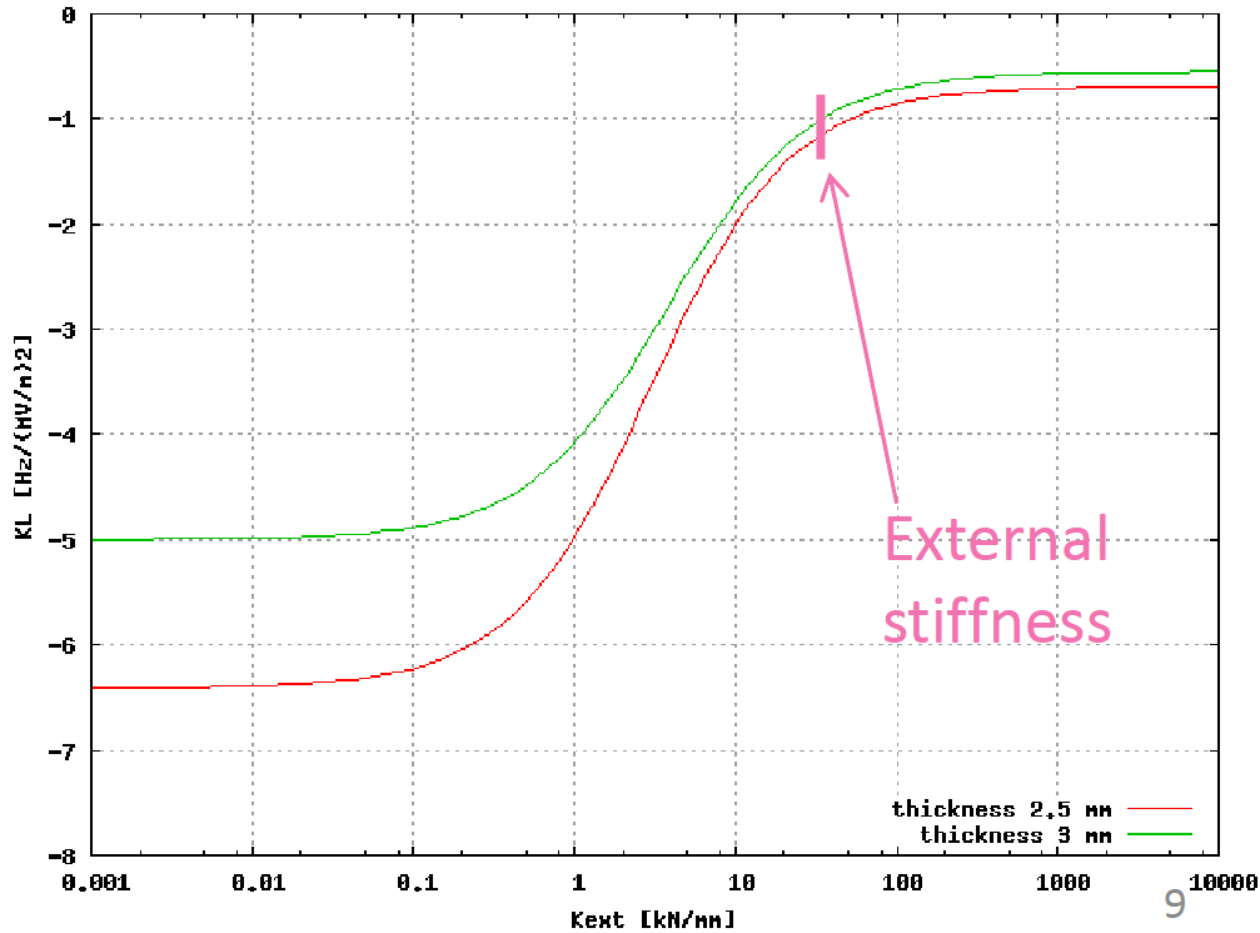


Helium tank

- Helium tank for short cryo-module cavities (first set)
 - Tank in Titanium for thermal contraction compatibility with respect to Niobium
 - Transition between Nb and Ti via NbTi to be EB welded on both sides
 - Flanges for external interfaces: HOM coupler, main coupler, inter-cavities, cryogenic lines in NbTi; alternative transition to cryogenic lines – bimetallic
- At least one stainless steel helium tank will be designed and manufactured in the frame of the 2nd set of cavities (2013) for the long cryo-module

Helium tank

- The stiffness of the helium tank has a direct impact on the Lorentz detuning (defines the boundary conditions of the cavity)



Juliette Plouin
CEA, Saclay

- Helium tank stiffness calculated by CEA

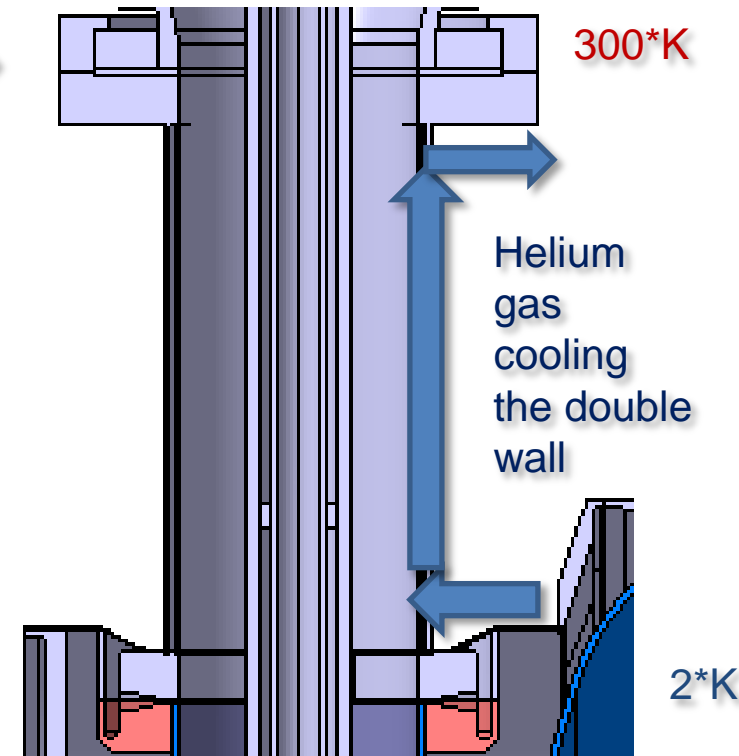
“Double wall” tube of main coupler

- Interface between the helium tank and the cryomodule vacuum vessel

- Parameters

- 1000 kW pulsed (100 kW average), 704.4 MHz, 50 Ω
- H=300 mm; D=100 mm; eint=1.5 mm; eext=2 mm
- Copper on stainless steel; Copper RRR = 30 (Sergio Calatroni)
- Cooling gas at 4.5 K input
- Lower part at 2K and upper part at 300K
- Heater at upper part to insure 30 °C of flange temperature

Heater 



“Double wall” tube of main coupler

- Model description (validated / the LHC main coupler)

- Copper on Stainless steel wall
- Semi-analytical model taking into account

- Conduction**

through the tube

$$Q_{\text{cond}}(T_a, T_b, i) := \int_{T_b}^{T_a} \frac{\lambda_{\text{ss}}(T) \cdot S_{\text{cond_tube_coupleur}}}{l_i} dT$$

- Convection**

$$Q_{\text{cv}}(T_{\text{wall}}, T_{\text{gas}}, h_c, S_{\text{convection}}) := h_c(T_{\text{gas}}) \cdot S_{\text{convection}} \cdot (T_{\text{wall}} - T_{\text{gas}})$$

- Radiation** between warm and cold parts

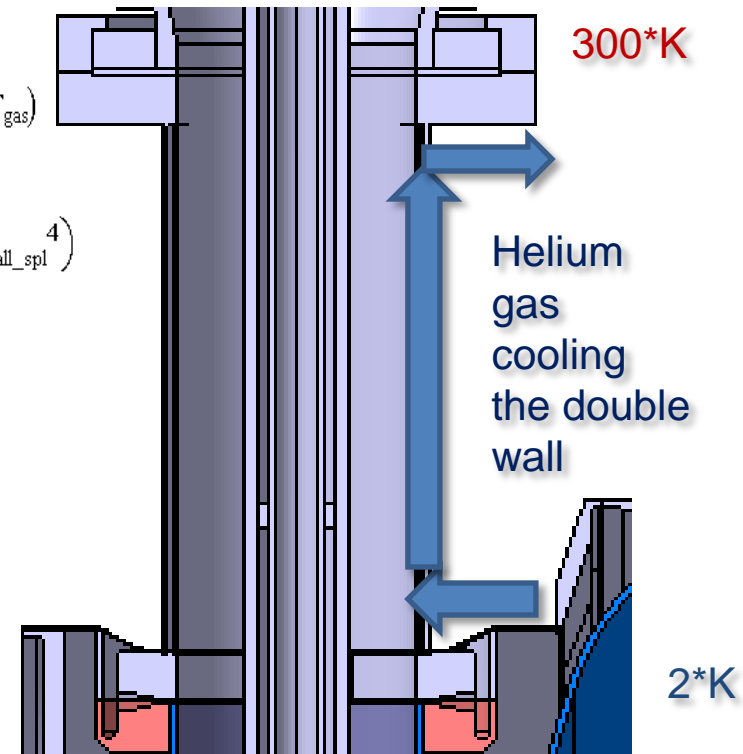
$$Q_{\text{rad_antenne_wall}} := A_{\text{wall}} \cdot \epsilon_{\text{antenne_wall}}(T_{\text{wall_spl}}) \cdot F_{\text{wall_antenne}} \cdot \sigma_z \cdot (T_{\text{antenne}}^4 - T_{\text{wall_spl}}^4)$$

- Power dissipation** (average) in the wall when coupler on

$$P_{\text{diss_ext}} := \int_0^{h_{\text{tube_coupleur}}} \frac{(I_{\text{peak_wall}}(x))^2}{2} \cdot R_{\text{ext_elect}}(\text{Temp}_{\text{tube_coupleur}}(x)) dx$$

$$I_{\text{peak_wall}}(x) := I_0 \cdot 2 \cdot \sin\left(\frac{\omega_0}{c} \cdot x\right) \quad I_0 := \sqrt{2 \cdot \frac{P_f}{Z_0}}$$

$$R_{\text{ext_elect}}(T) := \frac{\rho_{\text{CU_RRR30}}(T)}{\delta(\rho_{\text{CU_RRR30}}(T), \omega_0) \cdot \pi \cdot d_{\text{tube_coupleur_int}}}$$

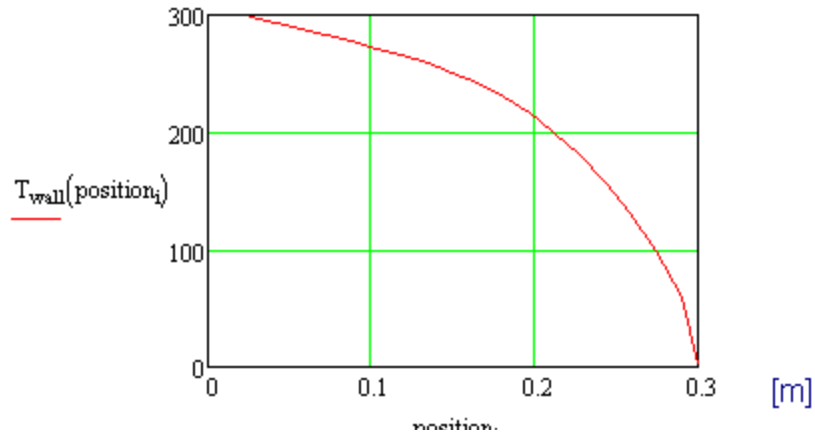


“Double wall” tube of main coupler

- Why cooling the wall?

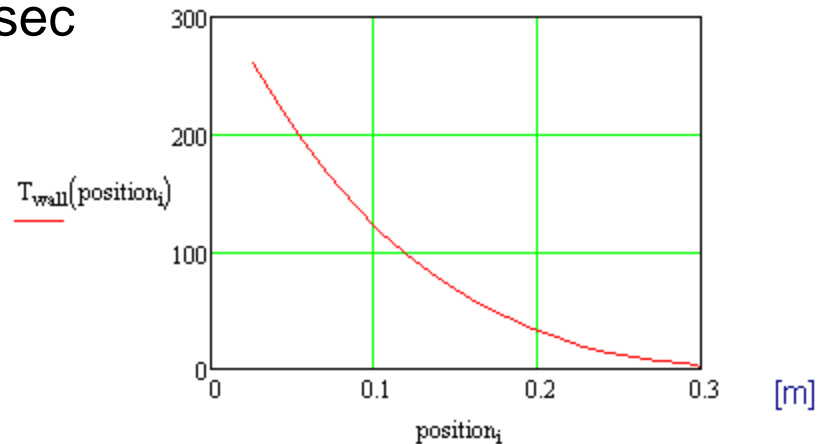
- No cooling temperature profile
=> Gives 21W to 2K

[K]



- Cooling with 42 mg/sec temperature profile
=> Gives 0.1W to 2K

[K]



“Double wall” tube of main coupler

- Why a heater at the top flange?
 - The heater insures 30 °C of flange temperature
 - If no heater, in order to have the same temperature at the flange when no power on
for the same thickness => height of more than 1m

“Double wall” tube of main coupler

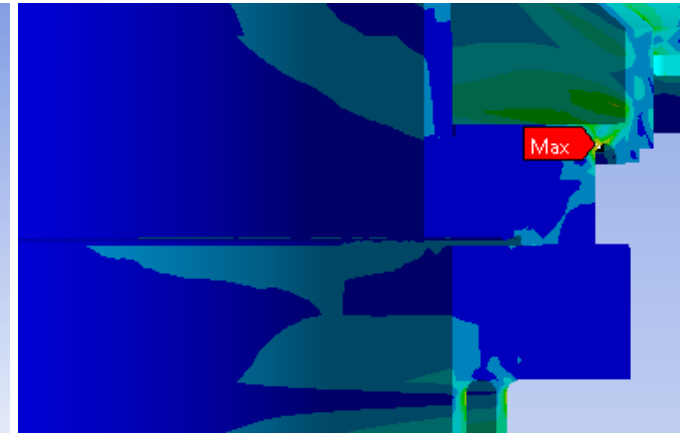
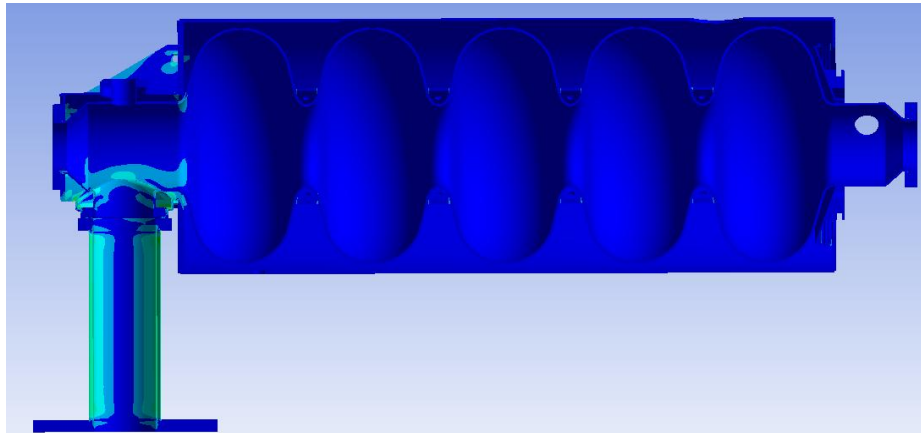
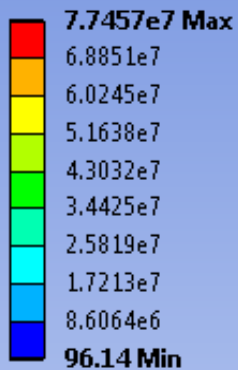
- Some thermal results

Massflow mgram/sec	21		23		28		35		42	
Power	ON	OFF	ON	OFF	ON	OFF	ON	OFF	ON	OFF
Temp. gas out	286 K	277 K	283 K	273 K	271 K	242 K	255 K	205 K	232 K	180 K
Q thermal load to 2K	2.4 W	0.1 W	1.7 W	0.1 W	0.4 W	0.1 W	0.1 W	0.1 W	0.1 W	0.1 W
Q heater	19 W	32 W	21 W	34 W	29 W	38 W	39 W	41 W	46 W	44 W
ΔL	0.1 mm (0.63-0.53)mm				0.05 mm (0.66-0.61)		~ 0 mm (0.67-0.67)			

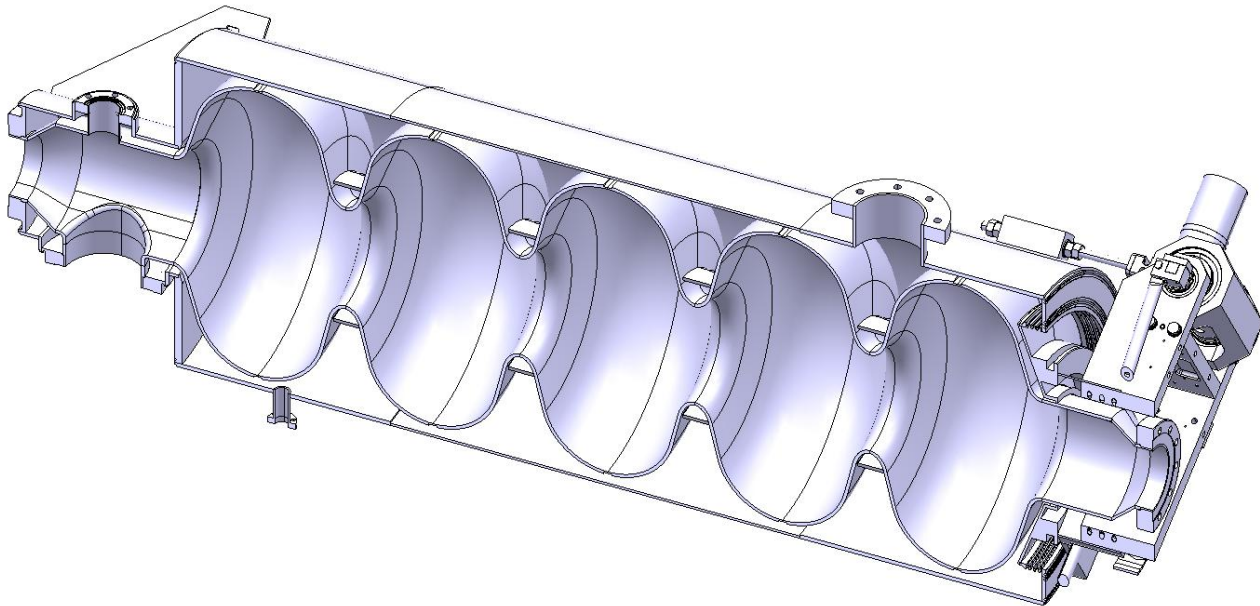
- Outlet tube for cooling gas inside the cryomodule
- Cryo people wish: remote controlled valve for massflow adjustment

“Double wall” tube of main coupler

- Mechanical considerations
 - Foreseen as supporting system for the cavity inside the cryomodule
 - Worst case: cavity supported in cantilever by the double wall => maximum stress in the double wall below the maximum allowable stress



- Helium tank / cryogenic lines
 - Operation at 2K with saturated superfluid helium
 - 2 connections to cryogenic lines
 - For initial fill-in – dimensioned for optimization initial fill time + instrumentation cables if any
 - For continuous cooling – detailed hereafter



- Some theory concerning Hell

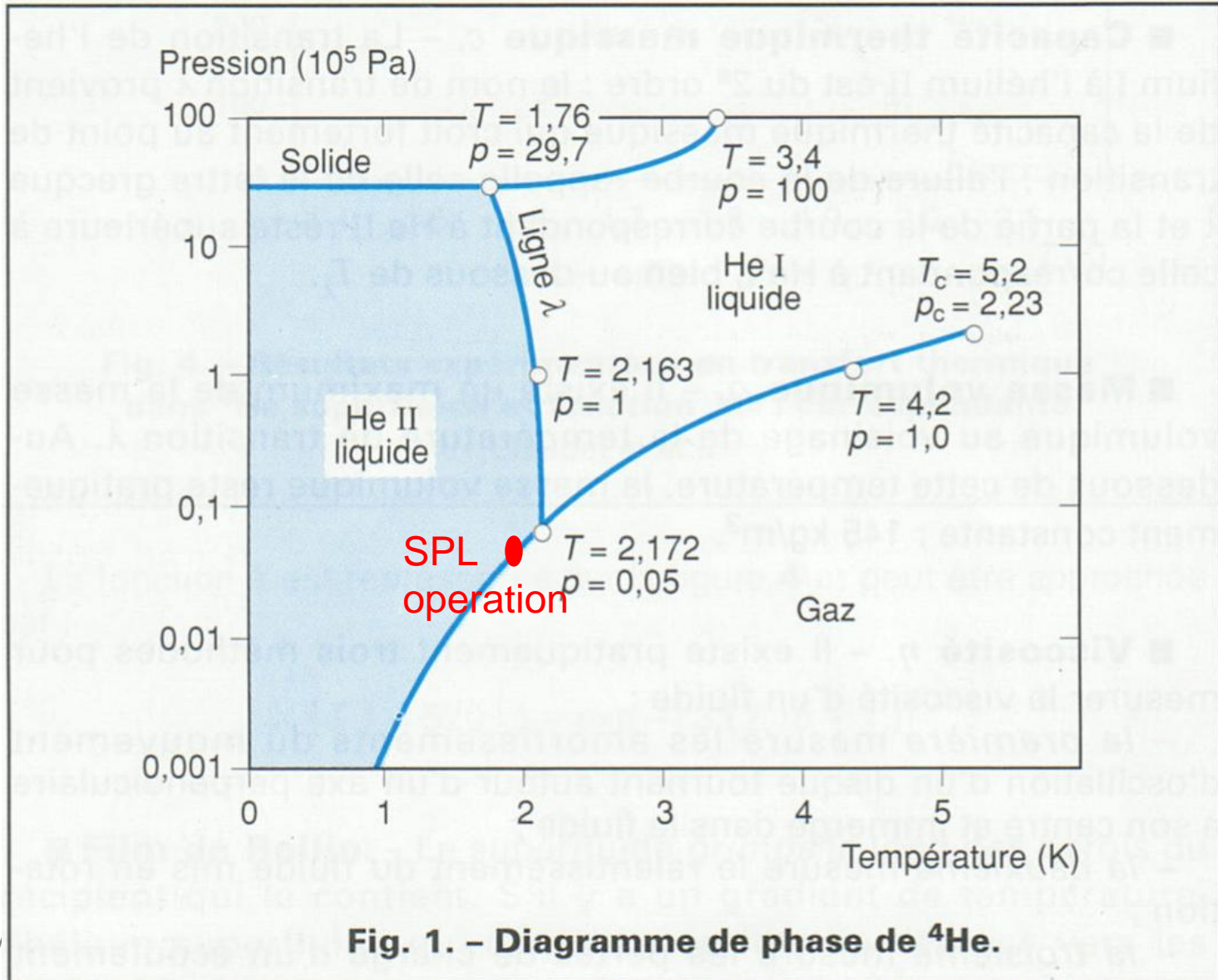
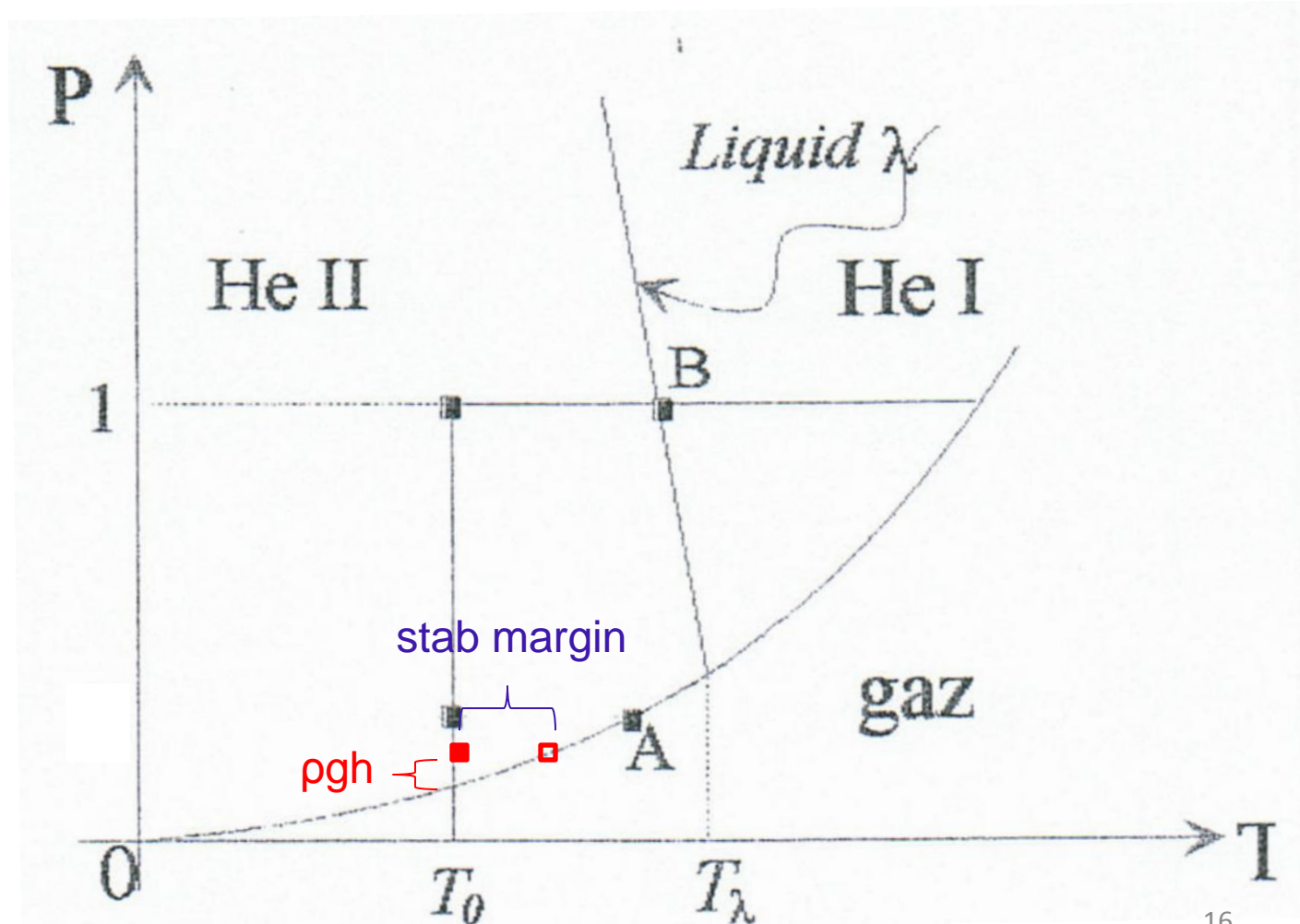
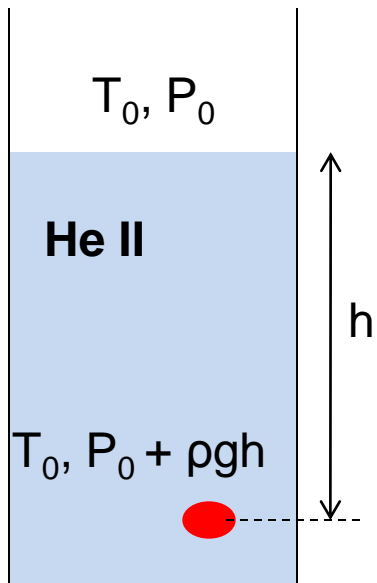


Fig. 1. - Diagramme de phase de ^4He .

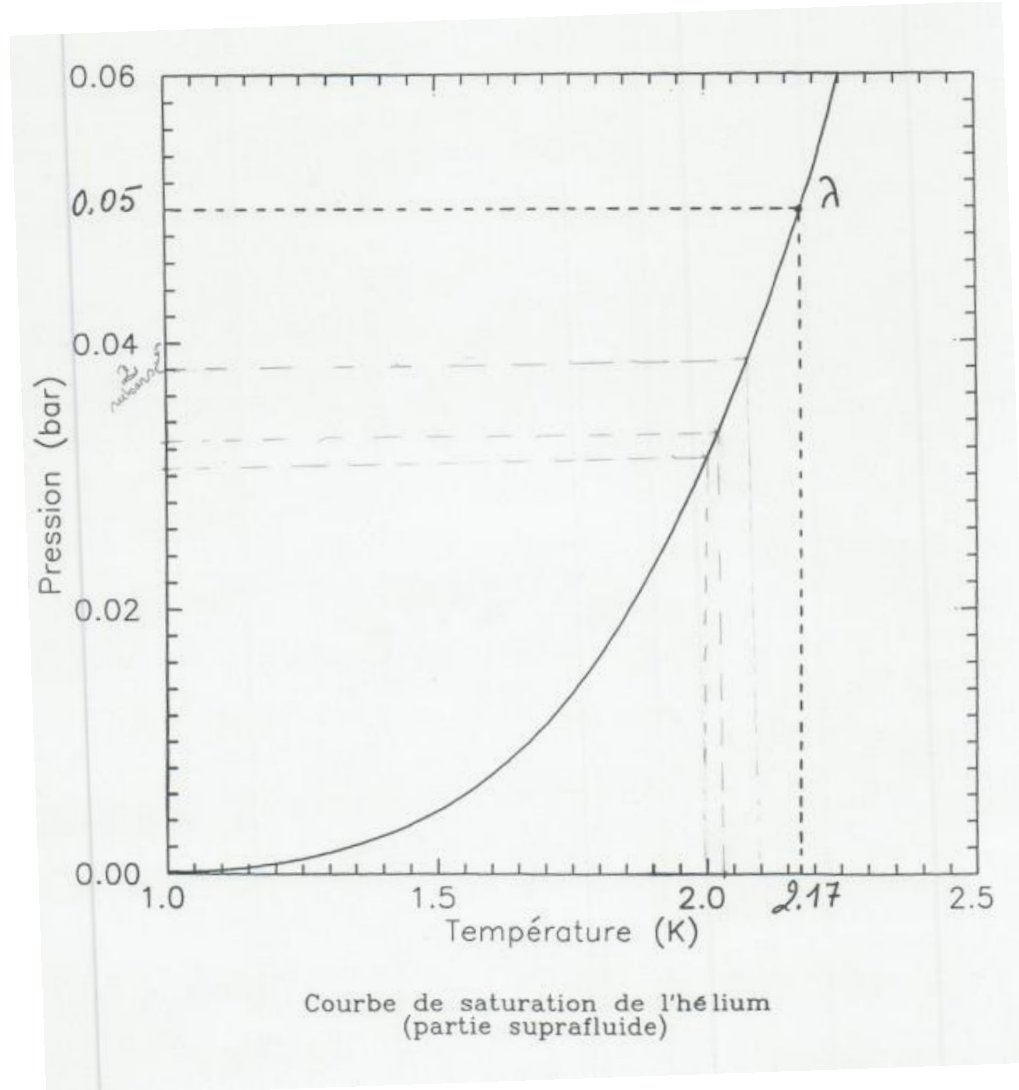
Interfaces to cryogenic lines

- HeII pressurized better stabilizer than HeII saturated
- Stabilization margin for saturated HeII due to hydrostatic pressure



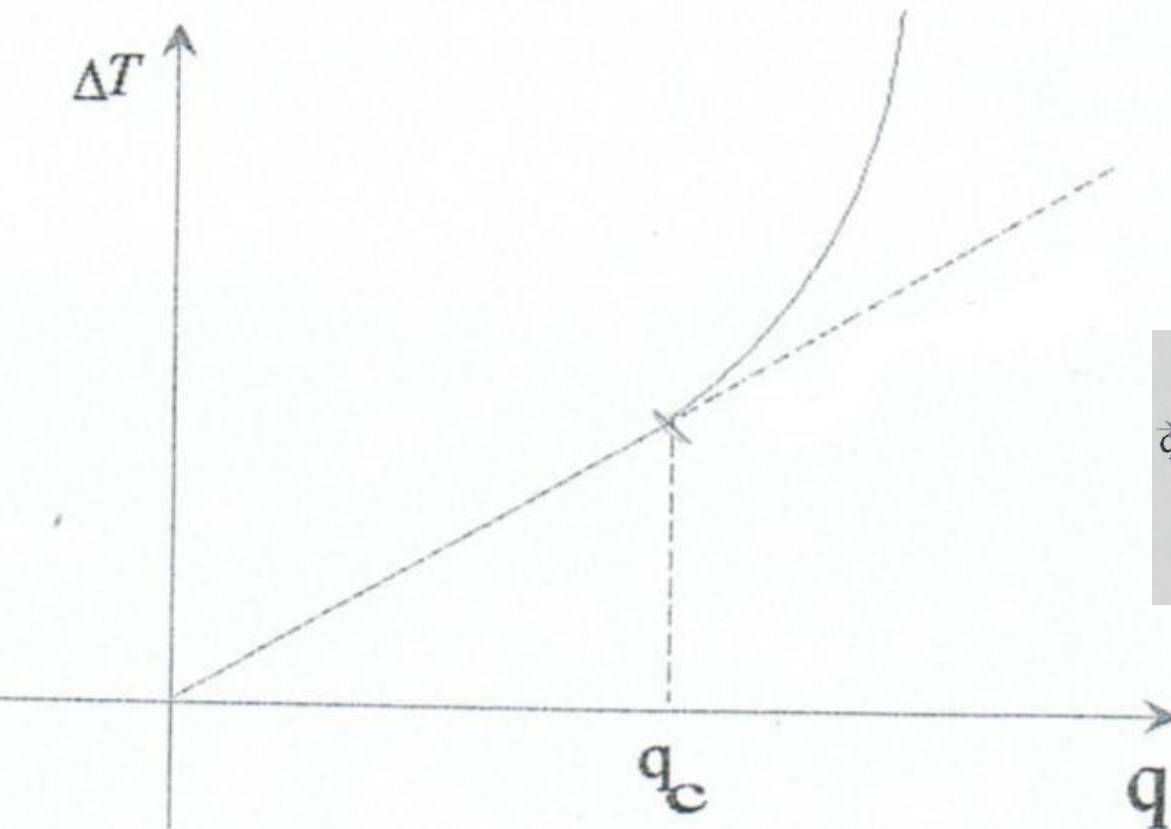
Interfaces to cryogenic lines

- For example, in saturated Hell, for a channel of $L=10$ cm height, $\Delta p = 1.4$ mbar, margin from 2K to ~ 2.025 K

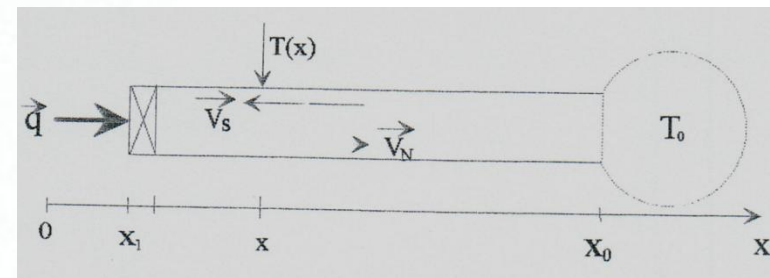


Interfaces to cryogenic lines

- Helium is an excellent thermal conductor; A typical value of “thermal conductivity” at 2K is 2kW/mK for a channel of cross section of 1 cm² and length 10cm (one order of magnitude higher than pure copper).
- However, this relationship is true only for small heat fluxes!!! Above a critical heat flux, the temperature increases drastically and eventually the superfluidity is lost.



The heat flux in Helium depends on temperature and channel dimension



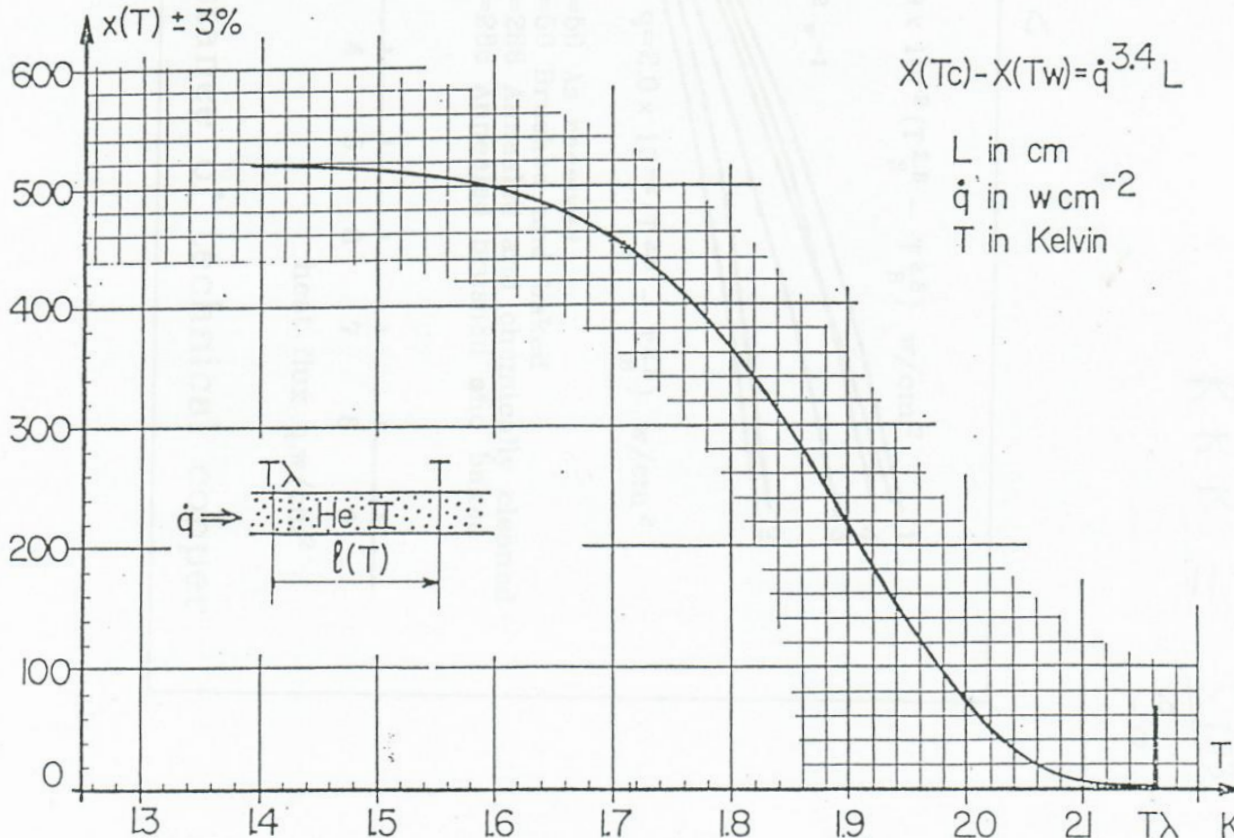
Interfaces to cryogenic lines

- Gorter and Mellinck have shown the dependence of the heat flux density / externally applied temperature
- Claudet et al. gave experimental values of heat transfer by Hell

$$\frac{\dot{Q}}{s} = \left[\frac{X(T_f) - X(T_c)}{l'} \right]^{0.29} \quad (\text{en W/cm}^2)$$

T_f et T_c (K) températures aux bouts froid et chaud d'un canal de section s (cm²) et de longueur l' (cm).

COURBE PRATIQUE



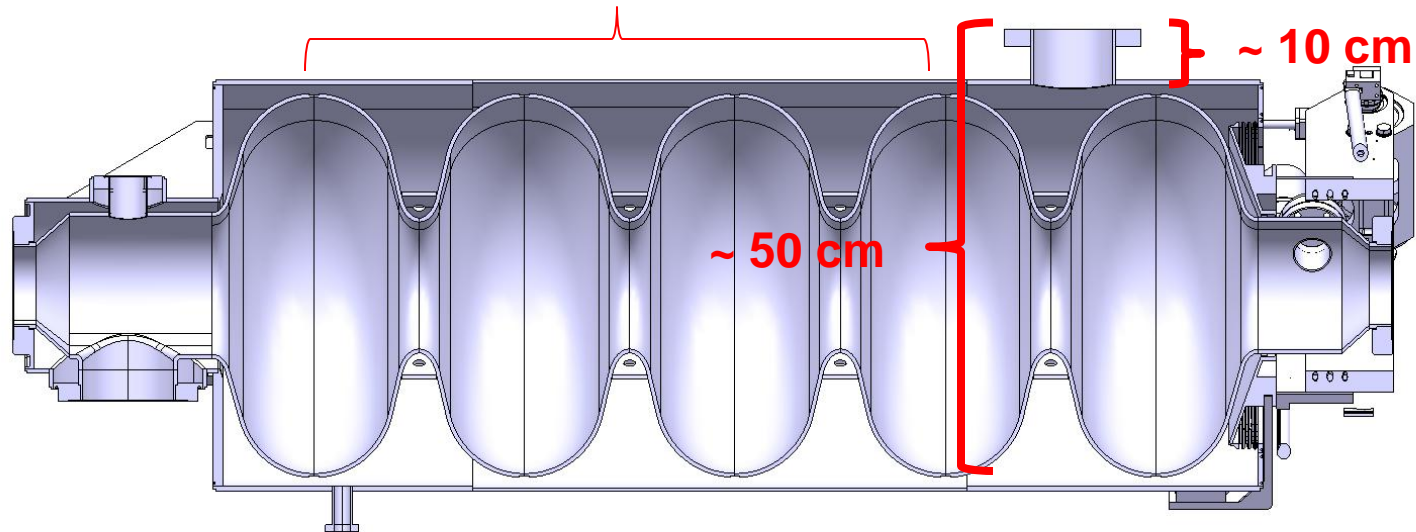
In our case:

$T_c = 2\text{K}$

$T_w = \text{temp stability margin (precedent example } 2.025\text{K)}$

Interfaces to cryogenic lines

- Application to our helium tank – some preliminary estimations
 - Heat dissipation mainly at the equator



- For the upper part $Q_c = \left[\frac{X(2K) - X(2.025K)}{10} \right]^{1/3.4}$
 - $Q_{c_up} \approx 1.5 \text{ W/cm}^2$
- For the lower part $Q_c = \left[\frac{X(2K) - X(2.1K)}{73} \right]^{1/3.4}$
 - $Q_{c_down} \approx 0.95 \text{ W/cm}^2$
- Left $Q_{c_left} \approx 0.8 \text{ W/cm}^2$

- Application to our helium tank – some preliminary estimations
 - Heat load per cavity, average: **16 W + static load from cryomodule + ? HOM discussed by Wolfgang**
 - Cryo duty cycle: 8.2 % => peak heat load **195 W + ... idem**
 - Dimensioning of the piping depending on the cavity test program:
 - To be able to extract 195 W continuously => cross section > **130 cm²** + optimization of space between cavity and helium tank
 - To be able to extract 25 W (margin as Vittorio) continuously => cross section > **17 cm²**
 - Rmq The total tank HeII temperature increase due to 195 W during 1.6 ms is only **10⁻⁵ K !!**

- Conclusions
 - Main coupler double wall optimized to withstand the induced mechanical loads and low thermal losses to the helium bath;
More details at <http://indico.cern.ch/materialDisplay.py?contribId=3&materialId=slides&confId=86123>
 - Interfaces of helium tank with the cryogenic lines to be dimensioned according to the maximum average heat load experienced by the cavity during its lifetime
 - Details of the helium tank will be discussed in a dedicated meeting